



Bellcomm

955 L'Enfant Plaza North, S.W.
Washington, D. C. 20024

date: December 6, 1971

to: Distribution

from: K. P. Klaasen

subject: Real-Time Determination of
PLSS O₂ Walkback Constraints
During Lunar Surface Traverses --
Case 310

B71 12007

ABSTRACT

PLSS O₂ is the limiting consumable for Apollo 16 lunar surface traverse distance and EVA duration. Pre-mission planning can keep the nominal traverses within the maximum return distance to the LM for which sufficient O₂ remains to walk back from a failed LRV. However, a method of keeping the actual traverses within this constraint during the mission is required to take into account variations between the planned and actual values for metabolic rates, initial PLSS charge, and O₂ leak rates. A series of boundary curves has been derived that can be laid over the real-time plot of O₂ quantity remaining versus EVA time to indicate the emergency walkback constraint. Using the actual O₂ usage rates and predicted time of O₂ depletion, the required departure time from any station can be easily determined. These boundary curves could be used by the TELMU team in Mission Control to inform the Experiment Officer of the stations that are time-limited by the O₂ walkback constraint well enough in advance to allow adequate real-time traverse planning.

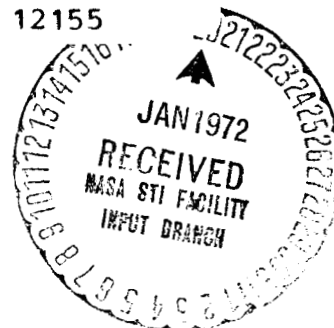
{NASA-CR-125365} REAL-TIME DETERMINATION OF
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MEMORANDUM FOR FILE

An analysis of the Apollo 16 lunar surface traverses shows the PLSS O₂ to be the limiting consumable on traverse distance and EVA duration. Figures 1-3 show the traverses plotted within their constraint envelopes for the three lunar surface EVA's. The traverses and traverse planning assumptions are those presented at the November 15, 1971 Science Working Panel Meeting. At both Station 8 on EVA 2 and Station 14 on EVA 3, the O₂ walkback limit is closely approached. Pre-mission planning can keep the nominal traverses within this O₂ constraint; however, such planning does not guarantee that the traverses as actually performed will remain within the constraint boundary. In fact, the O₂ boundary will shift for the actual traverses due to variations between the planned and actual values for metabolic rates, initial PLSS charge, and O₂ leak rates. Therefore, the O₂ walkback constraint as plotted in Figures 1-3 is of little use in real time. Some other method must be used during the Apollo 16 mission to insure that this constraint is not violated, and this method must take into account the effects of variations in metabolic rates, initial PLSS charge, and O₂ leak rates.

The O₂ walkback limit represents the maximum distance the crew can be from the LM at any given time during the EVA and still have enough O₂ remaining to be able to walk back to the LM from a failed LRV, ingress, and connect to the LM O₂ supply. The assumed walkback speed is 3.6 km/hr over uncorrected map distance for total return times of up to 1 hour and 2.7 km/hr for longer returns. The metabolic rates involved in an emergency walkback are assumed to be 1560 BTU/hr at 3.6 km/hr and 1290 BTU/hr at 2.7 km/hr. The PLSS O₂ must be sufficient to cover the walkback as well as 10 min of activity at the LRV to troubleshoot and/or gather equipment together and 13 min for LM ingress. The metabolic rate associated with this additional

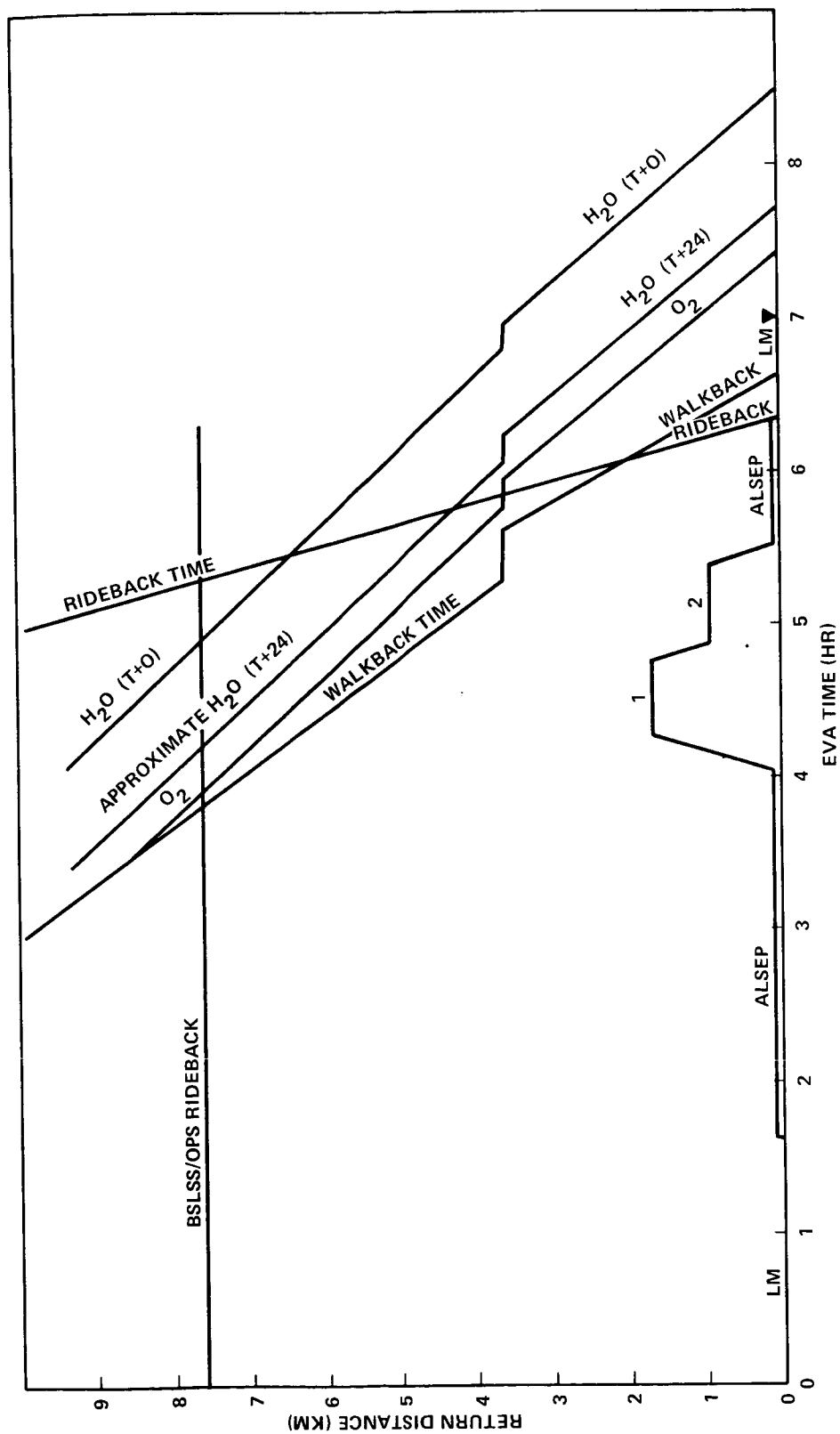


FIGURE 1 - APOLLO 16 LRV TRAVERSE CONSTRAINT ENVELOPE, EVA 1

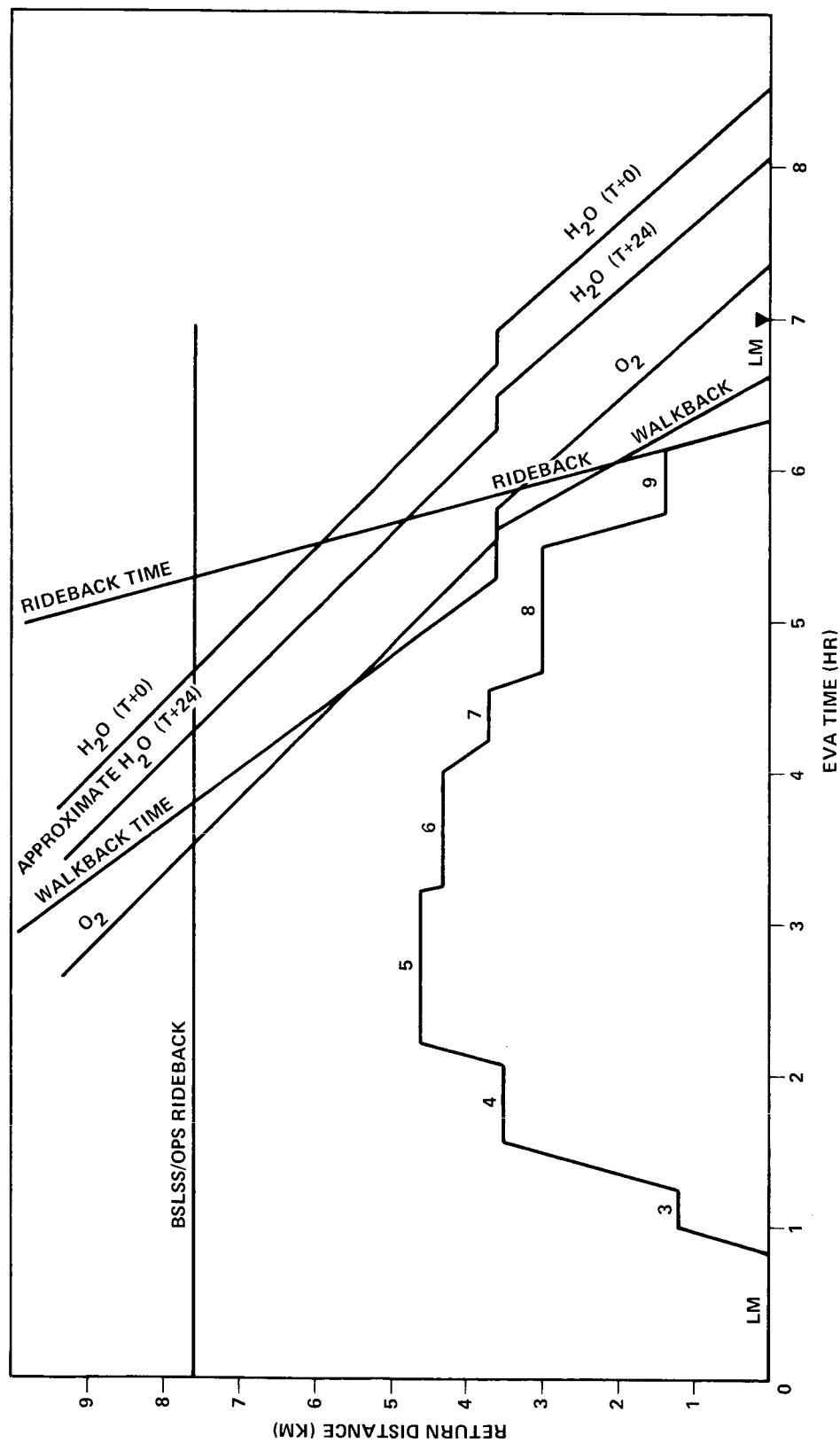


FIGURE 2 - APOLLO 16 LRV TRAVERSE CONSTRAINT ENVELOPE, EVA 2

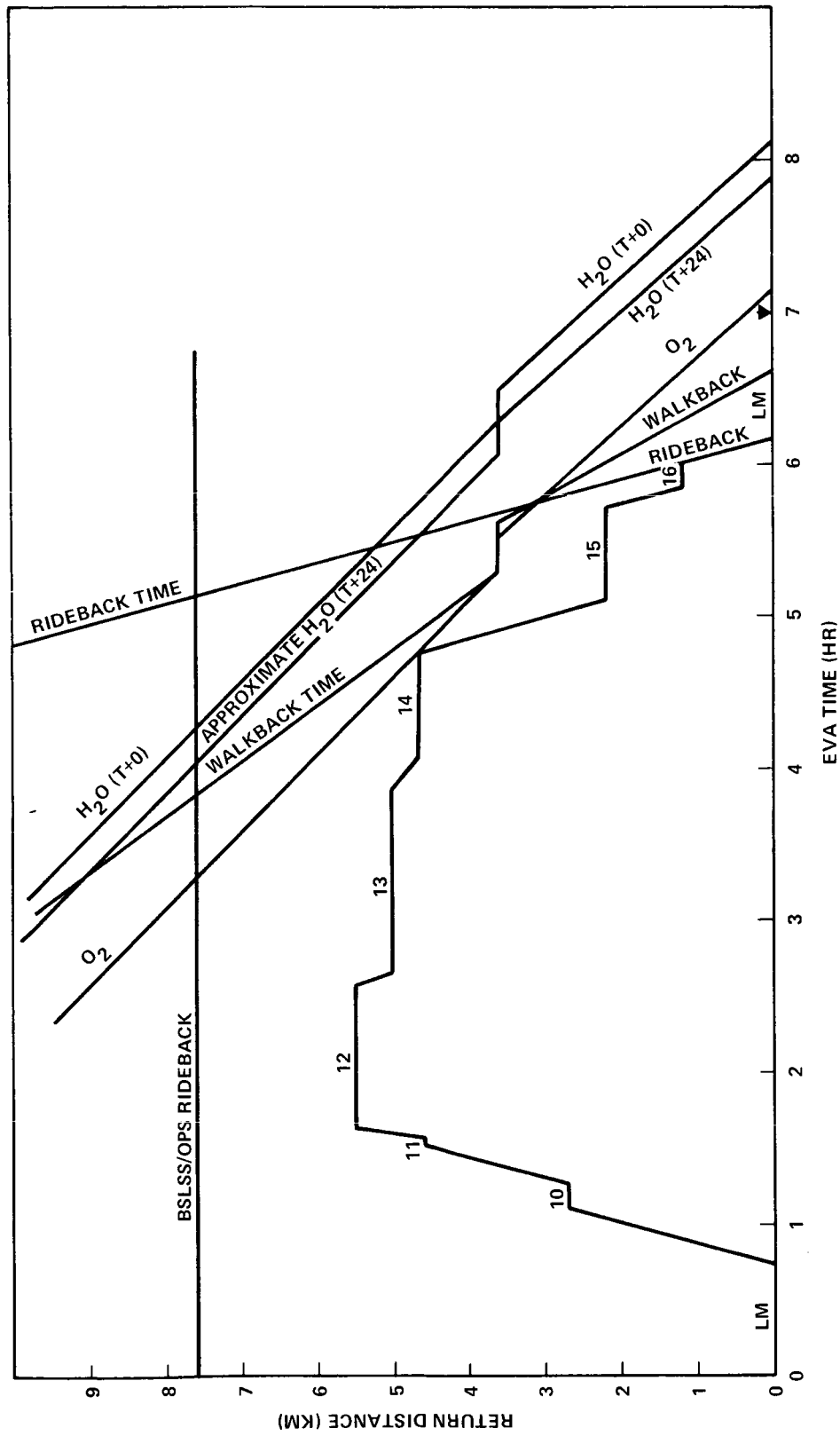


FIGURE 3 - APOLLO 16 LRV TRAVERSE CONSTRAINT ENVELOPE, EVA 3



- 2 -

23 min of activity is assumed to be the same as that for normal working. O_2 is consumed at a rate of

.0001627 (Metabolic Rate) + O_2 Leak Rate

in lb/hr. Therefore, for a traverse station a return distance D from the LM, a minimum quantity

$$Q = \frac{D}{R} (.0001627 M + L) + \frac{23}{60} \dot{O}_2$$

of O_2 would be required for an emergency walkback, where

Q = total quantity of usable O_2 remaining (lb)
 D = return distance to the LM (km)
 R = emergency walkback rate (km/hr)
 M = walkback metabolic rate (BTU/hr)
 L = O_2 leak rate (lb/hr)
 \dot{O}_2 = actual O_2 usage rate for working (lb/hr).

The crew must leave a station with at least Q lb of usable PLSS O_2 remaining to protect the emergency walkback requirement.

The actual quantity of PLSS O_2 remaining is tracked by the TELMU team in Mission Control during each EVA. The data are plotted on charts such as that of Figure 4. After about 3 to 3.5 hr of EVA, the O_2 consumption rates are extrapolated forward to predict the maximum duration of the EVA as far as O_2 quantity is concerned. Using the predicted time of usable O_2 depletion and the actual quantity of usable O_2 remaining at any point in the EVA, the minimum quantity of O_2 required to provide for an emergency walkback can be calculated as a function of return distance to the LM. If t is the time in hr remaining to depletion of usable O_2 , then \dot{O}_2 can be closely approximated by Q/t , and we have

$$Q = \frac{D}{R} (.0001627 M + L) + \frac{Q}{t} \left(\frac{23}{60} \right) .$$

Rearrangement gives

$$Q = \frac{Dt}{R(t - \frac{23}{60})} (.0001627 M + L) .$$

Plots of Q versus t for constant D can then be drawn using

$R = 3.6$ km/hr, $M = 1560$ BTU/hr for $D \leq 3.6$ km,
 $R = 2.7$ km/hr, $M = 1290$ BTU/hr for $D > 3.6$ km, and
 $L = .0275$ lb/hr.



The actual value of L for any EVA may differ from .0275 lb/hr, but pre-mission predictions range only from .02 to .035 lb/hr. Thus, the error in the calculation of Q using $L = .0275$ lb/hr is expected to be $< 3\%$. Figure 5 shows Q versus t for D ranging from 0 to 7 km. Interpolation between curves of constant D is linear for constant t . Two curves are plotted for $D = 3.6$ km. One curve is associated with walkback times greater than 1 hr, and the other with times less than 1 hr. Q becomes very large as t approaches 23 min in order to protect the time allotted at the LRV and for ingress. Q was calculated in lb and was converted to psia using the ratio of 800 psia/lb, which corresponds to a PLSS O_2 bottle temperature of about 85°F.

For use during a mission, a transparency of Figure 5 can be laid over a plot like Figure 4 to provide a quick and easy determination of the departure time from a station required to preserve the emergency walkback capability. The origin of Figure 5 should be located at the point at which usable O_2 depletion is predicted, as shown in Figure 6. A station at a return distance D from the LM must be departed by the time the actual O_2 quantity falls below the boundary in the overlay. For example, if O_2 consumption on EVA 3 of Apollo 16 is as it was on EVA 1 of Apollo 15, Figure 6 shows that Station 14 (4.65 km from the LM) would have to be departed about 2 hr 15 min before the predicted time of usable O_2 depletion. Since usable O_2 depletion is predicted at 6 hr 33 min, Station 14 would have to be departed at 4 hr 18 min into the EVA, or 28 min before the planned time, in order to protect the PLSS walkback capability.

The maximum expected error in the required departure time as determined using the curves of Figure 5 is about $(2 + 3t)$ min. Of this error, the 2 min is due to the maximum expected error introduced by approximating \dot{O}_2 by Q/t (corresponding to an error of 100 BTU/hr in the metabolic rate for working), and the $3t$ min is due to the maximum expected error in the assumed O_2 leak rate. The leak rate errors could be largely eliminated by constructing several families of boundary curves corresponding to a range of leak rates and using the set that matches the best estimate of the actual leak rate of the pacing crewman in any EVA. A somewhat more accurate value for \dot{O}_2 could also possibly be determined; however, the error associated with this value would not be much smaller than that involved in using Q/t .

Since several of the planned stations on the Apollo 16 traverses will likely be time limited by the O_2 walkback constraint, the TELMU team should be able to inform the Experiments Officer of the latest allowable departure time from such stops well enough



in advance to allow adequate real-time traverse planning. The proposed chart should make this possible with little difficulty and reasonable accuracy. Similar boundary curves could be constructed for PLSS H₂O walkback requirements; however, H₂O is not expected to be constraining for a fully charged PLSS.

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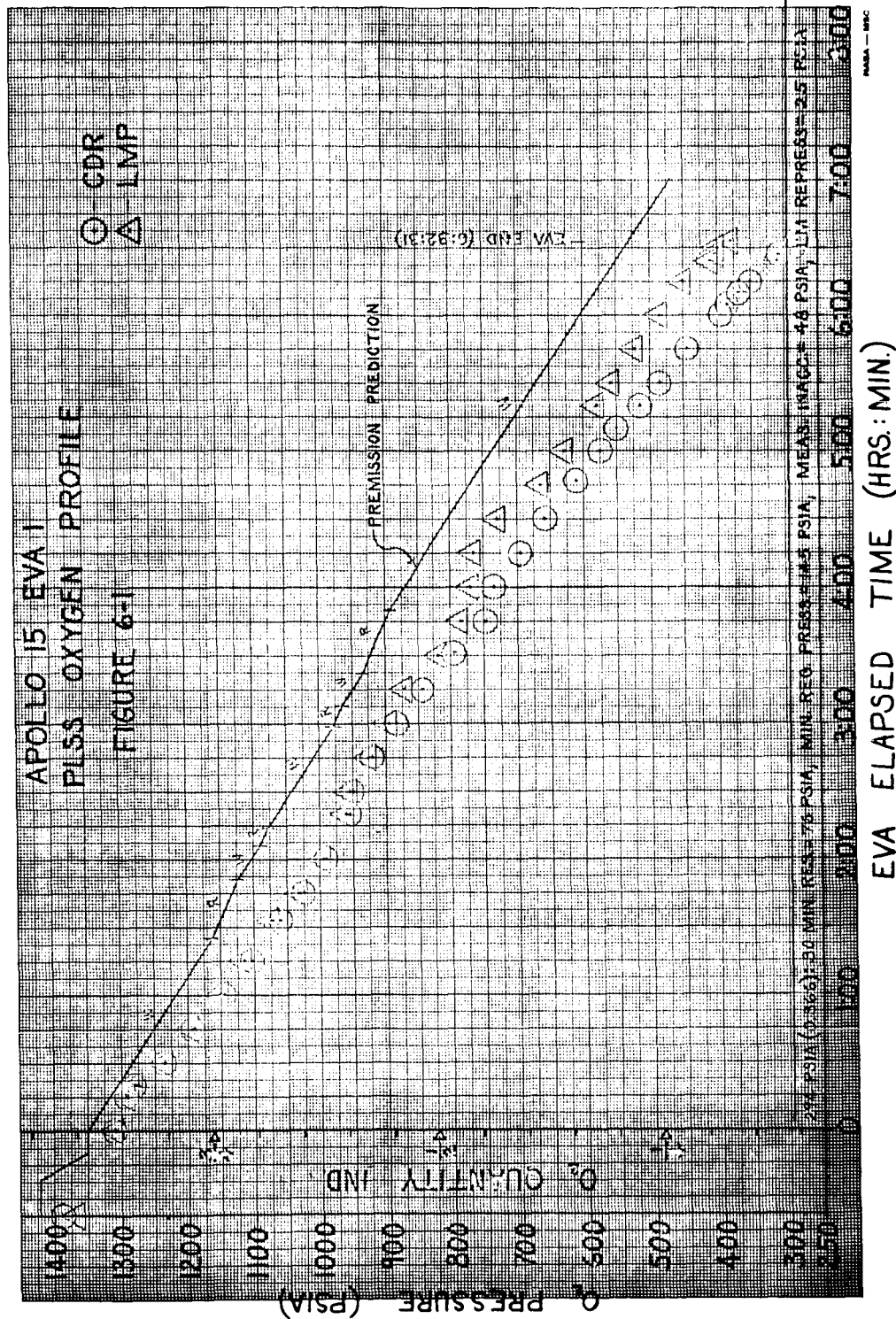


FIGURE 4 - REAL TIME PLOT OF PLSS O₂ REMAINING VS EVA TIME

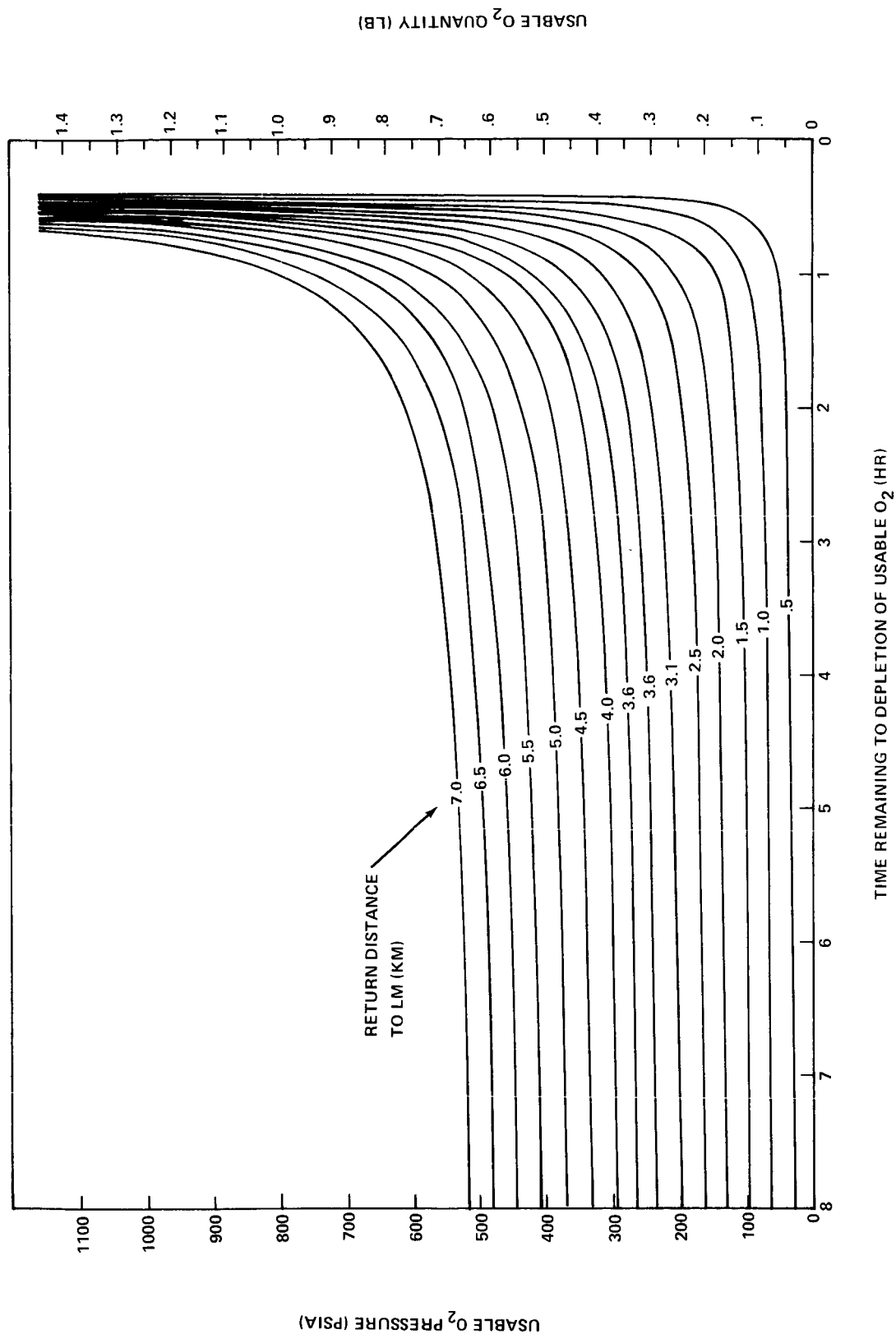
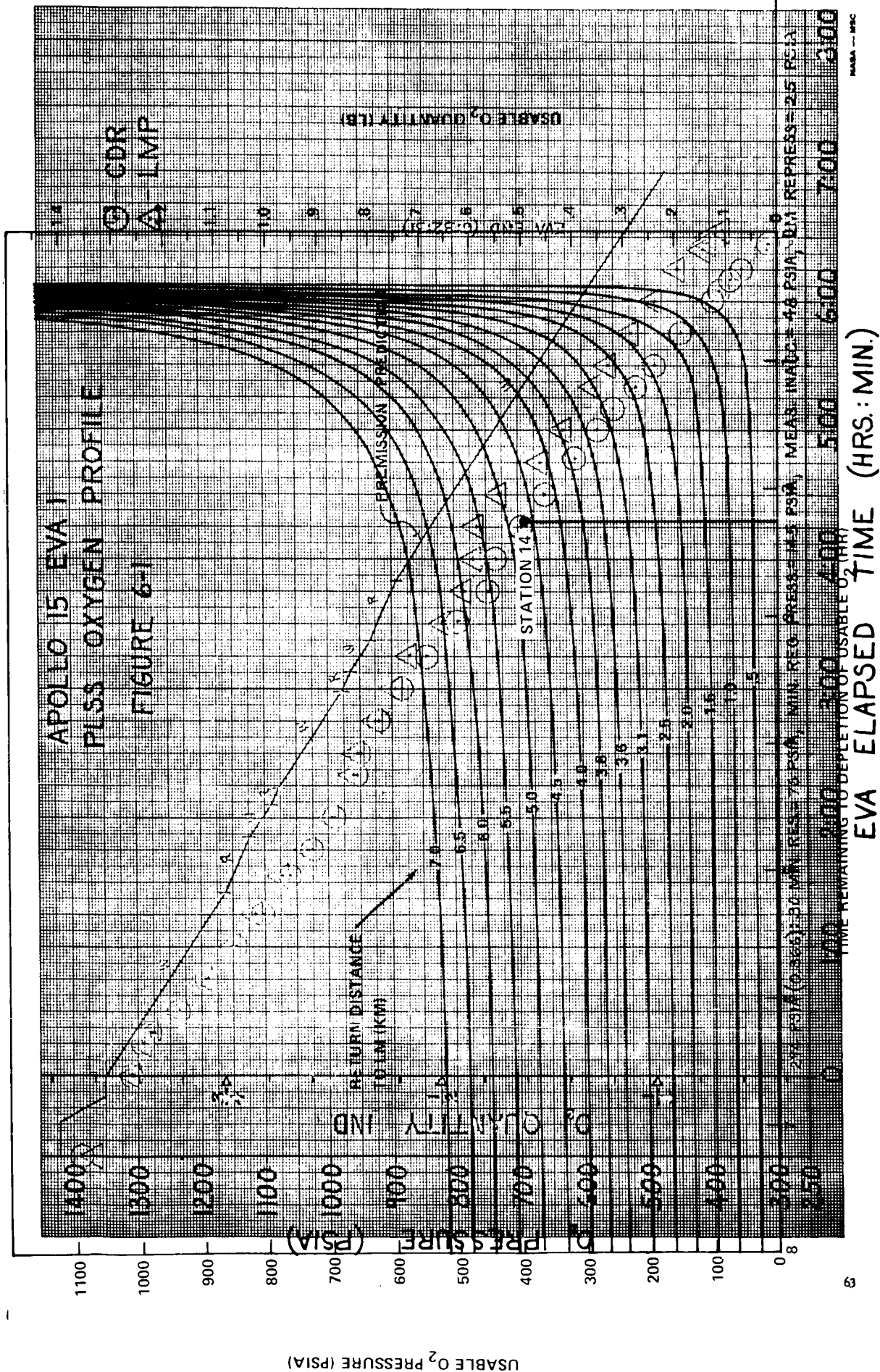


FIGURE 5 - REQUIRED QUANTITY OF USABLE O₂ VS PREDICTED TIME TO DEPLETION





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